

# **STARSHIP 1**

**SERIES 2000**

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**Beechcraft**  
A **Raytheon** Company

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## SPECIFICATIONS

### WEIGHTS

Maximum Ramp Weight	12,610 lbs.
Maximum Takeoff	12,500 lbs.
Maximum Landing Weight	11,875 lbs.
Maximum Zero Fuel Weight	10,800 lbs.
Basic Empty Weight (With Unusable Fuel, Oil and Standard Avionics)	8,011 lbs.
Useful Load (Standard Airplane)	4,599 lbs.

### WING AREA AND LOADINGS

Wing Area	280.88 sq. ft.
Wing Loading	44.5 lbs./sq. ft.
Power Loading (PT6A-67)	6.25 lb./shp.

### PRESSURIZATION

(8.4 Differential)	Cabin Altitude
* Actual Aircraft Altitude - 21,400 ft.	Sea Level
* Actual Aircraft Altitude - 25,000 ft.	1560 ft.
* Actual Aircraft Altitude - 30,000 ft.	3860 ft.
* Actual Aircraft Altitude - 35,000 ft.	5940 ft.
* Actual Aircraft Altitude - 41,000 ft.	8060 ft.
* Nominal Max Capability - Not Per Pre-Program	

## PERFORMANCE

**MAXIMUM SPEED** ..... 352 kts. (405 mph) (25,000 ft.)

### CRUISING SPEEDS (TAS) - AVERAGE CRUISE WEIGHT

#### MAXIMUM CRUISE POWER (1550 RPM)

At 20,000 ft.	350 kts. (403 MPH)
At 25,000 ft.	352 kts. (405 MPH)
At 30,000 ft.	350 kts. (403 MPH)
At 35,000 ft.	341 kts. (392 MPH)
At 39,000 ft.	325 kts. (374 MPH)

#### MAXIMUM RANGE POWER - (1550 RPM)

At 20,000 ft.	212 kts. (244 MPH)
At 25,000 ft.	225 kts. (259 MPH)
At 30,000 ft.	238 kts. (274 MPH)
At 35,000 ft.	255 kts. (294 MPH)
At 39,000 ft.	266 kts. (306 MPH)

### CRUISE RANGE FOR 508 GAL. (3400 LB) USEABLE FUEL

#### MAXIMUM CRUISE POWER (1550 RPM)

At 20,000 ft.	1166 nm. (1342 MI)
At 25,000 ft.	1361 nm. (1567 MI)
At 30,000 ft.	1610 nm. (1853 MI)
At 35,000 ft.	1915 nm. (2204 MI)
At 39,000 ft.	2257 nm. (2598 MI)

#### MAXIMUM RANGE POWER (1550 RPM)

At 20,000 ft.	1747 nm. (2011 MI)
At 25,000 ft.	2004 nm. (2307 MI)
At 30,000 ft.	2281 nm. (2625 MI)
At 35,000 ft.	2515 nm. (2895 MI)
At 39,000 ft.	2625 nm. (3021 MI)

### RATE OF CLIMB AT SEA LEVEL - TWO ENGINES

At 12,500 lbs. .... 3250 ft./min.

### DIMENSIONS

Aft Wing Span	54 ft.
Fwd Wing Span	
Cruise	20.92 ft.
Landing	23.96 ft.
Length	46 ft./1 in.
Height to Top of Vertical Stabilizer	12 ft./10 in.
Cockpit Height	58.18 in.
Cockpit Width	67 in.
Cockpit Door Width	18.5 in.
Cabin Length (Include Aft Baggage)	253.5 in.
(Excludes Pilot's Compartment)	
Cabin Width	66 in.
Cabin Height	65.5 in.
Entrance Door	28 x 50.39 in.
Electronic Compartment	
Volume	17.5 Actual, 13.4 Useable
Fwd Baggage Compartment	20 cu. ft./250 lbs.
Aft Baggage Compartment	35 cu. ft./385 lbs.
Total Baggage Capacity	55 cu. ft.

### RATE OF CLIMB AT SEA LEVEL - ONE ENGINE

At 12,500 lbs. .... 1180 ft./min.

### SERVICE CEILING - TWO ENGINES

(100 ft./min.) at 12,500 lbs. .... 39,900 ft.

### SERVICE CEILING - ONE ENGINE

(50 ft./min.) At 12,500 lbs. .... 29,400 ft.

### STALL SPEEDS (CAS)

Flaps 100%	79 kts. (91 MPH)
Flaps Up	94 kts. (108 MPH)

### TAKEOFF DISTANCE - FLAPS UP (12,500 lb.)

Rotation Speed (CAS)	97 kts.
Ground Run	1595 ft.
Total Distance Over 50 ft. Obstacle	2400 ft.

### LANDING DISTANCE

Approach Speed	103 kts.
Ground Roll	1990 ft.
Total Distance Over 50 ft. Obstacle	2880 ft.

### ACCELERATE - GO DISTANCE - 12,500 LBS.

Decision Speed (V <sub>1</sub> )	93 kts.
Obstacle Speed (V <sub>2</sub> )	106 kts.
Total Distance Over 35 ft. Obstacle	2960 ft.

### ACCELERATE - STOP DISTANCE - 12,500 LBS

Decision Speed (CAS)	93 kts.
Total Distance	2990 ft.

# STANDARD EQUIPMENT

## AVIONICS/FUNCTIONS

<b>COMM No. 1</b>	Collins VHF Communications 20 Watt Transceiver with Frequency Range from 118.000 to 136.975 MHZ.
<b>COMM No. 2</b>	Collins VHF Communications 20 Watt Transceiver with Frequency Range from 118.000 to 136.975 MHZ.
<b>NAV No. 1</b>	Collins Navigation Receiver provides VOR/Localizer from 108.000 to 117.95 MHZ in 50 KHZ increments, 40 Channel Glideslope Receiver and Marker Receiver.
<b>NAV No. 2</b>	Collins Navigation Receiver provides VOR/Localizer from 108.000 to 117.95 MHZ in 50 KHZ increments, 40 Channel Glideslope Receiver and Marker Receiver.
<b>ADF</b>	Collins Single ADF Receiver provides ANT, ADF and Tone Modes. Tunes the range from 190 KHZ to 1749.5 KHZ in 0.5 KHZ steps. It also tunes the 2181 KHZ Emergency Frequency.
<b>DME</b>	Collins Dual DME Receivers providing distance up to 300 nautical miles, rate, closure time and station identifier.
<b>Transponder Nos. 1 &amp; 2</b>	Collins Mode A/C Transponder Units with 4096 Codes and altitude reporting capability. Ident Button - Pilot and Copilot Control Wheel.
<b>Radio Altimeter</b>	Collins Radio Altimeter System provides height up to 2500 feet and is displayed on Pilot and Copilot's Panel.
<b>Radio Tuning</b>	Collins Radio Tuning Units provide the Pilots with central displays of the VHF COMM, VHF NAV, ATC and ADF Frequencies and the easy ability to change both the frequencies and modes of operation. A Concentric Knob Tuning Control provides easy operation in turbulence. Dual controls with access to both the Pilot and Copilot's radios provide full redundancy.
<b>Compass System No. 1</b>	Collins Strap-down Reference System providing attitude and heading measurements as well as angular rates and linear accelerations in the three aircraft axes. Provides heading and attitude to the Pilot's Flight Director Instruments and Autopilot.
<b>Compass System No. 2</b>	Collins Strap-down Reference System providing attitude and heading measurements as well as angular rates and linear accelerations in the three aircraft axes. Provides heading and attitude to the Copilot's Flight Director Instruments and Autopilot. Provides heading to the Pilot's Sensor Display Unit.

**Weather Radar** Collins Solid-State Weather Radar with Turbulence Detection capability displayed on the Pilot and Copilot's Navigation and Multifunction Displays of the Electronic Flight Instrument Systems. Incorporates a dual channel Weather Radar with Control Functions to independently control the dual display capability which makes it appear to the Pilots as if there were two Radars in the aircraft.

**Pilot's EFIS/EICAS Displays** An Electronic Multicolor Display System consisting of four display units with integral symbol generators is provided. These displays provide all the primary flight, navigation, engine and system presentations, including conventional functions for Attitude Director Indicator, Horizontal Situation Indicator and Engine Instruments. In addition, these displays provide for the display of map data, weather radar, flight mode annunciation, flight path information, checklist and caution/warning messages. Reversary capability is provided for EFIS and EICAS data.

**Copilot's EFIS Flight Director System** Two (2) Primary Flight and Navigation Displays identical to those of the Pilot tied to the No. 2 Attitude-Heading Reference System. Provides Copilot with Independent Flight Director System. Autopilot has capability to couple to either Flight Director System.

**Flight Management** A Flight Management System provides automatic point-to-point geographic-based navigation using VOR/DME sensor inputs plus signals from a Long-Range Navigation Sensor, position computation in the FMS is based on latitude and longitude irrespective of the Position Fixing Sensor used. Thus the system is capable of worldwide point-to-point, great circle navigation at all times so long as sensors are available suitable to the particular location. An extensive built-in non-volatile memory data base is included, which can be updated on board the aircraft, VNAV capability is included and either CDU can be used to keyboard tune all of the radios if desired by the pilot.

<b>Air Data</b>	<p>Dual Digital Air Data Computers for the Flight Control Systems calculate the Air Data Parameters, True Airspeed and Encoded Altitude Information.</p> <p>Pilot's displays include two (2) 4 x 4 Multicolor CRT's. One provides Mach Airspeed, Predicted Airspeed, Digital Display of True Airspeed and Standard Air Temperature. The second CRT display provides Analog and Digital Altitude and Vertical Speed Displays, along with Digital Display of Selected Altitude and Baro-Correction.</p> <p>Copilot's displays include two (2) 4 x 4 Multicolor CRT's which provide identical information as Pilot's CRT Air Data Displays.</p>
<b>Autopilot</b>	<p>A Digital Fail Passive Autopilot System with integrated mode select and engage panel offering minimum workload. The system provides enhanced conventional Flight Control Systems Modes plus additional modes designed to increase performance without burdening the pilot.</p>
<b>Sensor Display Unit</b>	<p>Pilot's Monochromatic CRT Sensor Display which provides heading and VOR/FMS/ADF Bearing, also provides secondary display of NAV information.</p>
<b>Altitude Awareness</b>	<p>Provides control for the selection of decision height, minimum decision altitude and reporting altitude.</p>
<b>Audio</b>	<p>Dual Audio System to provide voice and transmission from all receivers and transmitters in the aircraft. Provides interphone between Pilot and Copilot, aural warnings to the cockpit and announcements from cockpit over cabin speaker system.</p>
<b>Standby Horizon</b>	<p>2-Inch Standby Gyro Horizon with Battery Pack.</p>
<b>Standby Airspeed</b>	<p>2-Inch Airspeed Indicator.</p>
<b>Standby Altimeter</b>	<p>2-Inch Counter-Pointer Altimeter.</p>
<b>System includes:</b>	<p>All necessary antennas  Mike Key Buttons Pilot and Copilot's Wheel  Line Advance Buttons Pilot and Copilot's Wheel  Dual Cockpit Speakers  Dual Hand-Held Mikes  Dual Boom/Mic/Headsets  Avionics Master Switch  Ground Clearance Switch tied to Comm No. 1 and Pilot's Audio</p>

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## LANDING GEAR AND BRAKES

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Tricycle Type Steerable Nose Wheel equipped with shimmy damper, Beech oil-air struts designed for smooth taxiing.

Dual Main Wheel Tires (each side)  
Standard Wheels and Tires - H19.5 x 6.75-10-6 ply rated  
Nose Wheel Tire - 19.5 x 6.75-8-10 ply rated  
Two Disc Brakes on Each Main Wheel  
Landing Gear Position Lights, Down and Locked  
Landing Gear Warning Horn and In-Transit Light

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## ENGINES

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Two Pratt & Whitney PT6A-67 Free Turbine Engines flat rated at 1000 Shaft Horsepower each  
Propellers - 100" Diameter, Four Blade Composite, Full Feathering, Hydraulically Controlled Constant Speed  
Fuel Crossfeed System  
Submerged Electric Standby Fuel Boost Pumps  
Jet Type Fuel Transfer Pumps  
Low Fuel Quantity Warning System  
Engine Driven Fuel Pumps  
Engine Driven Fuel Boost Pumps  
Fuel Control Units  
Primary Propeller Governors  
Overspeed Propeller Governors  
Fuel Topping Governors  
Automatic Fuel Heater System  
Complete Engine Anti-Icing System with Bleed Air Heated Engine Inlet  
Engine Fire Detection System  
Magnetic Chip Detector  
Auto Ignition System  
EPA Fuel Drain Collection System

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## CONTROLS

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Dual Conventional 3-Axis Control System  
Dual Adjustable Rudder Pedals and Toe-Operated Brakes  
Parking Brakes with Hand Control  
Two Power Levers, two Propeller rpm Selectors and two Cut-Off and Condition Levers  
Hydraulic Landing Gear  
Hand-Pump Type Auxiliary Landing Gear Extension Control  
Elevon Trim Tabs (Pitch and Roll)  
Electric Rudder Trim Tabs  
Yaw Damper System  
Electric Flap Controls

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## FLIGHT PACKAGE

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PRIMARY FLIGHT DISPLAY (PFD) -  
Presents a forward view of flight situation.  
Pitch and Bank Attitude  
Slip  
Flight Director Steering  
Flight Guidance/autopilot Mode/Status  
Glideslope Deviation  
Lateral Deviation (VOR, LOC, FMC)  
Marker Beacon  
Radio Altitude and DH  
MDA and Reporting Altitude Readout

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# STANDARD EQUIPMENT

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## FLIGHT PACKAGE (continued)

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### NAVIGATION DISPLAY (ND) -

Presents a plan view situation in a tactical time frame.

Heading (MAG)

Lateral Deviation (Localizer, VOR, FMS)

VNAV Vertical Deviation

Bearing (VOR, ADF, FMS Waypoint)

Distance (DME, FMS Waypoint)

Ground Speed (DME, FMS)

IDENT (DME, Waypoint)

Time-to-Go (DME, FMS Waypoint)

Flight Plan Waypoint (Labeled with Identifier)

Drift Angle

Wind Vector

Selected Categories of MAP Data

Weather Radar Display Data

Weather Radar Mode/Status Data

Select Active and Preset Courses

FMS Desired Track

Selected Heading

Compass Slaving Indication

VNAV Distance to Path/Waypoint

VNAV Waypoint Altitude

VNAV V/S Required

### ALTIMETER/VERTICAL SPEED (ALI)

Barometrically-derived Altitude - (Conventional full-size circular display with large numerical readout-rotating Drum.)

Altitude Preselector/Alerter

Reference "Bug" Function on Vertical Speed Scale

### ALTITUDE AWARENESS PANEL

Select Decision Height

Select Minimum Decision Altitude

Select Reporting Altitude

### AIRSPEED DISPLAY (ASI) -

Conventional Circular display with digital presentation.

Conventional - Calculated Airspeed

- Maximum Allowable Airspeed

- Airspeed Trend

- Airspeed Reference Bug

Digital Readout - True Airspeed (TAS)

- Static Air Temp (SAT) with Momentary Total Air Temperature (TAT)

### SENSOR DISPLAY UNIT (SDU)

- Basic RMI function with bearing pointer from ADF or VOR or FMS

### CONTROL AND DISPLAY UNIT (CDU) - Keyboard Controller for:

PFD/ND Modes and Data Selection

Weather Radar

Radio Tuning

Data Input/Display for FMS

Long Range NAV Sensor Control and Status Display

### ENGINE INDICATION AND CREW ALERTING SYSTEM (EICAS)

Engine Related Readouts:

Torque

Prop Speed Synchronization

ITT

Oil Temp

N<sub>1</sub>

Oil Pressure

Propeller Speed

Fuel Flow

Crew Alerting Readouts:

Caution/Advisory

### MULTIFUNCTION DISPLAY (MFD)

Weather Display

Checklist

FMS Formatted Map

FMS Tabular Formats

Avionics Fault Status and Diagnostic Formats

Backup Display for ND/EICAS

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## ELECTRICAL

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Starter Generator (Two 300 amp - 28 volt)

Solid State Generator Control Panel - Two

4 Electric Motors for Operating Flaps

1 Electric Motor for Forward Wing

Landing Gear Warning System with Manual Over-Ride and Automatic Reset

Heated Stall Warning System with Preflight Self Test System

Dual Heated Pitot & Static Heads - Nose Mounted

External Power Receptacle with Annunciation and Overvoltage Protection

Automatic Solid State Master Warning and Annunciators, with Self Test and Dimmer System (integrated into EICAS)

Heated Fuel Vents

Static Wicks

Air Cooled NiCad Battery - 34 AH

Triple Busing System with Cockpit Checkout Capability/Auto Load Shedding

Nose Gear Mic and Phone Jack (part of Avionics)

Two Loadmeters

One Voltmeter

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## LIGHTS

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Wing Ice Light

Two Landing Lights

Nose Gear Taxi Light

Flush Position Lights

Dual Map Lights

Adjustable Reading Light for each Cabin Chair

Indirect Cabin Lighting (with Passenger Dimming Feature)

Two Cockpit Overhead Reading Lights

Cabin Door Inspection Lights

Aisle Courtesy Light

Entrance Door Light

Aft Compartment Lights

Primary and Secondary Instrument Lighting Systems

Lighting Controlled from Overhead Panel

EL Indirect Cockpit Lighting

Flush Anti-Collision Strobe System (high and low intensity selectable)

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## COCKPIT

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Electrically Heated, Safety Plate Glass Windshields with Redundant Pilot Element & Control  
Hot Air Windshield Defroster  
Fail Safe (Dual Pane) Side Cockpit Windows - Pilot's Side Window Defrost  
Dual Adjustable Sun Visors  
Map Pocket  
Cigarette Lighter, Two Ash Trays  
Fresh Air Outlets  
Oxygen Outlets and Console Mounted Diluter Demand Masks w/Mic Sub Panel Mounted Oxygen Controls  
Coffee Cup Holders  
Dual Cockpit Speakers  
Pilot and Copilot 4-Way Adjustable Seats with Shoulder Harness Restraint System and Reclining Feature  
POH Storage Container  
Two Fuel Quantity Indicators  
One Anti-Ice Fluid Quantity Indicator (Time Remaining)

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## SERVICE

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Tow Bar  
Service Information Kit  
Two Pitot Tube Covers  
Sump Drain Wrench  
Two Engine Log Books  
Airplane Log Book  
Six Coat Hangers  
Pilot's Check List  
Power Chart  
Flight Bag  
Battery Manual  
Beechcraft Warranty ID Card

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## CABIN

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Fresh Air Outlets for All Occupants  
Air Conditioning (Standard)  
Ash Trays for All Occupants  
Automatic Bleed Air Type Heating and Vapor Cycle Cooling System with High Capacity Ventilation System  
Fail Safe (Dual Pane) Cabin Windows with Shades

"No Smoking - Fasten Seat Belt" Signs with Audible Chime  
Wall-to-Wall Carpet  
Six Cabin Chairs, Fully Adjustable, Swivel and Shoulder Harness, Lap Belt, and Retractable Inboard and Outboard Arm Rest  
Cupholder for Each Cabin Chair  
Forward Partition with Doors Separating Entry Way from Cabin and Cockpit  
Forward Luggage and Coat Closet  
Aft Partition with Door Separating Baggage Compartment from Cabin  
Private Lavatory with Relief Tube  
Emergency Exit  
Pressurization - 8.4 Differential  
Oxygen System (77 cu. ft.) complete with 1 Automatic Deployment Mask per Passenger  
Airstair Door with Hydraulic Snubber  
Airstair Door Courtesy Light  
Occupants Briefing Cards  
Emergency Exit  
Forward and Aft Cabin Blower  
4 Cabin Speakers

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## SPECIAL FEATURES

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Capacitance Type High Accuracy Temperature Compensating Fuel Gaging System  
Fuel Filter Anti-Syphon Valves  
Weather Avoidance Radar  
Inertial Separator Engine Anti-Icing System, with Dual Electric Ice Vane Actuators  
Anti-Icing Equipment  
Automatic Fuel Transfer System  
Complete Exterior Urethane Paint  
Dual Bleed Air Heating and Pressurization  
Dual Pitot and Static Systems  
Electroluminescent Lighted Control Panels  
508 Gallon Usable Fuel  
Battery Charging Current Sensor (Safety Sentinel)  
Emergency Locator Transmitter  
Pilot's Chartholder  
External Oxygen Filler Ports and Pressure Gage

**Beechcraft**

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**Wichita, Kansas 67201, U.S.A.**

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